

A finite element model using a unified formulation for the analysis of viscoelastic sandwich laminates

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1 Introduction

Sandwich plates with viscoelastic core are very effective in reducing and controlling vibration response of lightweight and flexible structures, where the soft core is strongly deformed in shear, due to the adjacent stiff layers. The theoretical work on constrained layer damping can be traced to DiTaranto [1] and Mead and Markus [2] for the axial and bending vibration of sandwich beams. Since then, different formulations and techniques have been reported for modelling and predicting the energy dissipation of the viscoelastic core layer in a vibrating passive constrained layer damping structure [3–5]. Other proposed formulations in-

clude thickness deformation of the core layer [6] and deal with the cases where only a portion of the base structure receives treatment [7].

Due to the high shear developed inside the core of the sandwich, equivalent single layer plate theories, even those based on higher order deformations, are not adequate to describe the behaviour of these sandwiches, also due to the high deformation discontinuities that arise at the interfaces between the viscoelastic core material and the surrounding elastic constraining layers. The usual approach to analyse the dynamic response of sandwich plates uses a layered scheme of plate and brick elements with nodal linkage. This approach leads to a time consuming spatial modelling task. To overcome these difficulties, the layerwise theory has been considered for constrained viscoelastic treatments, and most recently, Moreira et al. [8,9], among others, presented generalized layerwise formulations in this scope.

A review and assessment of various theories for modeling sandwich composites with application to sandwich beams can be found in the work of Hu et al. [10].

More recently, Araújo et al. [11–14] have presented and used for optimisation and viscoelastic material identification purposes a sandwich finite element model based on an eight noded serendipity plate element. The viscoelastic core layer is modelled according to a higher order shear deformation theory and adjacent elastic and piezoelectric layers are modelled using the first order shear deformation theory. All materials are considered to be orthotropic, with elastic layers being formulated as laminated composite plies. Passive damping is accounted for by using the complex modulus approach, allowing for frequency dependent viscoelastic materials and active damping is incorporated through feedback control laws for co-located control. Also in this framework, Moita et al. [15] developed a simple and efficient non conforming triangular finite element where the viscoelastic core is modelled according to Reissner-Mindlin laminated plate theory

and the face layers are modelled according the Kirchhoff-Love plate theory. Another sandwich plate model presented by Moita et al. [16] is based on Reddy's third order shear deformation theory for the core and the face layers are also modelled according to the classical laminated plate theory. These models also contemplate hybrid active-passive damping. A similar model was also presented by Bilasse et al. [17] for non linear vibrations of sandwich plates.

In the present work the stiffness and mass matrices are obtained by Carrera's Unified Formulation (CUF), firstly proposed in [18–20] for laminated plates and shells and extended to functionally graded (FG) plates in [21–23]. The present formulation considers a displacement-based layerwise formulation, with linear expansion of displacements in each layer, with degrees of freedom u_x, u_y, u_z at each lamina interface. To the authors knowledge, it is the first time that CUF is applied to this class of problems with viscoelastic behaviour. The main advantage of the present formulation is its versatility allowing for great flexibility in the through the thickness approximations, which is an important feature in sandwich structures.

The dynamic response of the finite element model is validated using a few reference solutions from the literature.

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